Standard Test Methods for Interlaminar Shear

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Introduction

Delamination in composites is a key failure mode, reducing structural integrity and life-span. Standard testing is needed to understand delamination and design composites resistant to failure. Due to the COVID-19 pandemic, planned physical testing moved towards attempting numerical modelling as a means of determining the shear properties of a unidirectional carbon fibre/ epoxy laminate. Investigating changes in interlaminar shear strength (ILSS) as laminate thickness increases.

Results

- Linear relationship between load, displacement and ILSS
- Laminate acts completely linearly elastic at all deflections
- No failure mechanisms displayed within data:
- **1.** No load drops from delamination/ matrix cracking/ fibre-matrix debonding.
- 2. No changes in stress distribution at high deflection where the model would be dominated by compression between the supports, rather than shear.
- **Overall: Modelling highlights Solidworks FEA shell modelling is** ineffective at determining ILSS and its corresponding failure mechanisms.





Solidworks Finite Element Analysis (FEA)

- (8-12)
- **Tested in 3-point flexure (maximising shear stress)**
- Loaded until beam deflection = 0.05, 0.1, 0.15, 0.2, 0.25 mm
- Extracted values for ILSS (MPa) and Load (N) at each deflection



Solidworks FEA: 8 ply model deflection plot (0.25mm)

Advanced composite modelling methods

- Advanced modelling is required to model the distinct failure modes occurring at the interface between plies, where matrix cracking, fibre-matrix debonding and ultimately delamination occurs. By accurately modelling how the initiation and propagation of these failure mechanisms occur, ILSS can be calculated. Advanced methods researched:
- Advanced interface elements
- 2. Micro scale modelling
- 3. 3D FEA modelling
- **Conclusion: A mix of interface modelling, 3D FEA, micro scale modelling and classical laminate** theory may be used to model interlaminar shear failure and calculate ILSS at an accuracy comparable to that of physical testing.

2D shell model (orthotropic-homogenous laminate properties) with ply numbers